

Verification of Welded eaves moment connection of open sections

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1 Description

The objective of this study: verification of CBFEM IDEARS software with component method. Description of verified connection: welded moment frame, horizontal beam is welded on the column flange, column stiffened with two horizontal stiffeners in levels of the beam flanges. Compressed parts are designed as maximal 3rd class to avoid stability problems (horizontal stiffeners of column, web panel in shear or compression, compressed beam flange). Horizontal beam is considered as both sides fixed beam of length 6m loaded by continuous load over the entire length=>vertical shear force and bending moment in the plane are of the same absolute values.

2 Analytical model

Component method

Six components is examined: fillet weld, web panel in shear, column web in transverse compression, column web in transverse tension, column flange in bending and beam flange in compression.

All components designed according to EN 1993-1-8.

Design loads of component depend on the position:

Web panel in shear– design loads on the vertical axis of the column

Other components- reduced design loads in column flange to which is connected horizontal beam.

3.1 Fillet weld

The weld is closed around a cross-section of the beam.

The thickness of the weld on the flanges can differ from the thickness of the weld on the web.

Vertical shear force is transferred only by welds on the web and plastic stress distribution is considered. Bending moment is transferred by whole weld shape and elastic stress distribution is considered.

Effective weld width depending on the horizontal stiffness of the column is considered (because of bending of the column flange).

Design of the weld is done according to EN1993-1-8 – 4.5.3.2(6).

The assessment is carried out in two major points: on the upper or lower edge of the flange (maximum bending stress) and in the crossing of the flange and the web (combination of shear force and bending moment stresses).

3.2 Web panel in shear

The thickness of the column web is designed to be maximally third class to avoid stability problem see EN1993-1-8 – 6.2.6.1(1).

Two contributions to the load capacity are considered: resistance of the column wall in shear and the contribution from the frame behaviour of the column flanges and horizontal stiffeners see EN1993-1-8 – 6.2.6.1(6.7 and 6.8).

3.3 Column web in transverse compression

Effect of the interaction of the shear load is considered see EN1993-1-8 – 6.2.6.2(tab. 6.3).

Influence of longitudinal stress in the wall of the column is considered see EN1993-1-8 – 6.2.6.2(2).

Horizontal stiffeners prevent from stability problem

The horizontal stiffeners are included in the load capacity of this component with the effective area.

3.4 Column web in transverse tension

Effect of the interaction of the shear load is considered see EN1993-1-8 – 6.2.6.2(tab. 6.3).

The horizontal stiffeners are included in the load capacity of this component with the effective area.

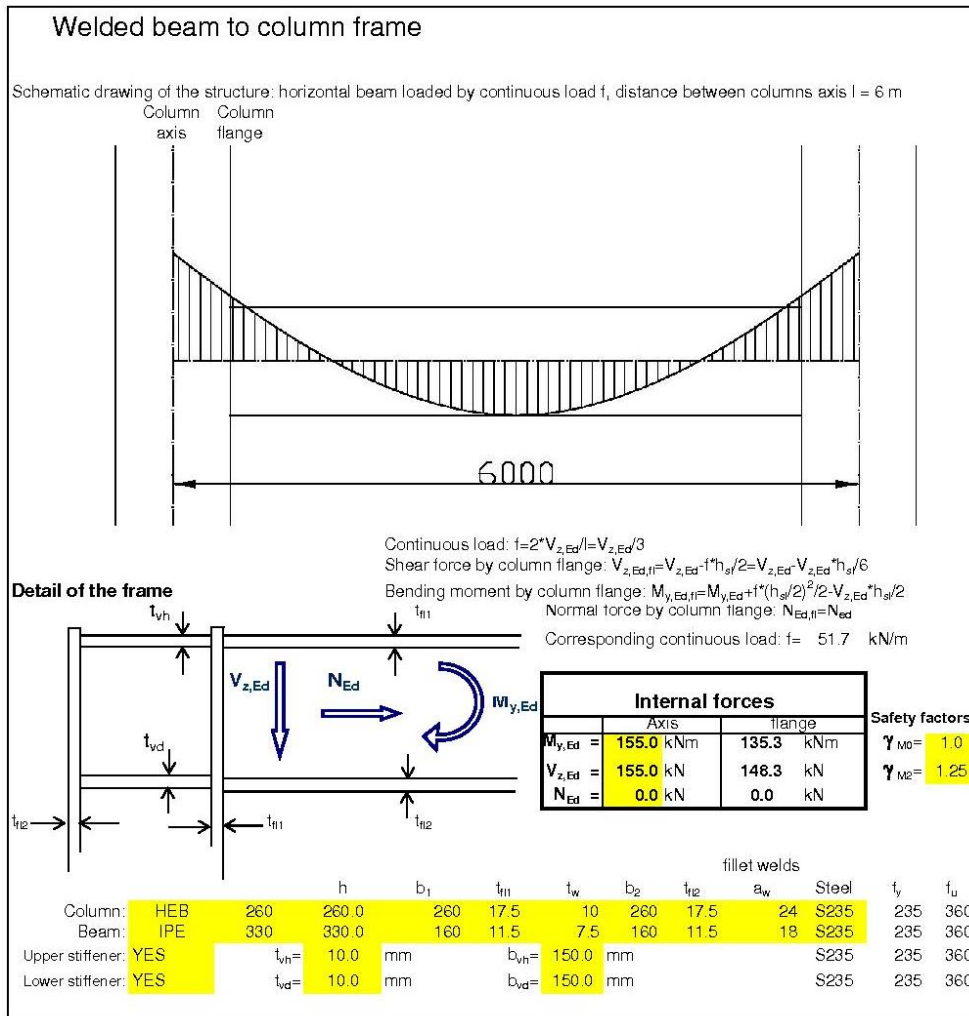
3.5 Column flange in bending

Horizontal stiffeners brace column flange, this component is not considered.

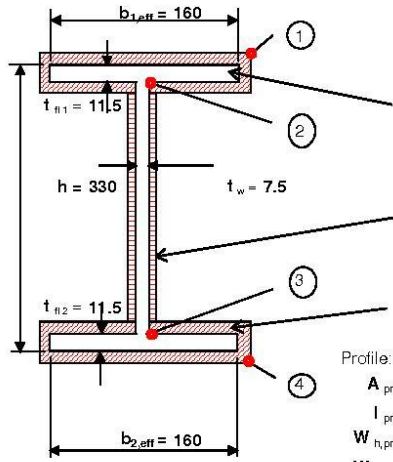
3.6 Beam flange in compression

The horizontal beam is designed to be maximally third class to avoid stability problem.

For better understanding of the component method design of the beam IPE330 to column HEB260 connection is shown below



All assessments according to EN 1993-1-8



$h = 330$ mm
 $b_{1,eff} = 160$ mm
 $t_{n1} = 11.5$ mm
 $a_{w,n1} = 6$ mm
 $t_w = 7.5$ mm
 $a_{w,w} = 5$ mm

$b_{2,eff} = 160$ mm
 $t_{n2} = 11.5$ mm
 $a_{w,n2} = 6$ mm

Profile:
 $A_{prof} = 5983$ mm²
 $I_{prof} = 111451454$ mm⁴
 $W_{t,prof} = 675463$ mm³
 $W_{d,prof} = 675463$ mm³

$\alpha = 1.0$

$M_{y,Ed} = 135.3$ kNm
 $V_{z,Ed} = 148.3$ kN
 $N_{Ed} = 0.0$ kN

$f_{y,k} = 235$ MPa
 $f_u = 360$ MPa
 $\gamma_{M2} = 1.25$
 $\beta_w = 0.8$

Distance of critical points to the center of gravity:

$z_1 = 171.0$ mm
 $z_2 = 153.5$ mm
 $z_3 = 153.5$ mm
 $z_4 = 171.0$ mm

Welds:
 $A_w = 6700$ mm²
 $I_w = 117044933$ mm⁴

Resistance of the weld

4.5.3.2(6) Considered internal forces: by column flange

Point ①

$$\sigma_{\perp} = \tau_{\perp} = \frac{M_{y,d} z_1 / (I_w \sqrt{2}) + N_d / (A_w \sqrt{2})}{\text{Utilization}} = \frac{139.8 \text{ MPa}}{0.54} < \frac{0.9 f_u}{\gamma_{M2}} = 259.2 \text{ MPa} \quad \text{OK}$$

$$\sqrt{\sigma_{\perp}^2 + 3 \tau_{\perp}^2} = 279.5 \text{ MPa} < \frac{f_u}{\beta_w \gamma_{M2}} = 360.0 \text{ MPa} \quad \text{OK}$$

Point ②

$$\tau_{\parallel} = \frac{V_{z,d} / (2a_{w,w} (h - t_{n1} - t_{n2}))}{\text{Utilization}} = \frac{46.3 \text{ MPa}}{0.48} < \frac{0.9 f_u}{\gamma_{M2}} = 259.2 \text{ MPa} \quad \text{OK}$$

$$\sigma_{\perp} = \tau_{\perp} = \frac{M_{y,d} z_2 / (I_w \sqrt{2}) + N_d / (A_w \sqrt{2})}{\text{Utilization}} = \frac{125.5 \text{ MPa}}{0.48} < \frac{0.9 f_u}{\gamma_{M2}} = 259.2 \text{ MPa} \quad \text{OK}$$

$$\sqrt{\sigma_{\perp}^2 + 3(\tau_{\perp}^2 + \tau_{\parallel}^2)} = 264.5 \text{ MPa} < \frac{f_u}{\beta_w \gamma_{M2}} = 360.0 \text{ MPa} \quad \text{OK}$$

Point ③

$$\tau_{\parallel} = \frac{V_{z,d} / (2a_{w,w} (h - t_{n1} - t_{n2}))}{\text{Utilization}} = \frac{46.3 \text{ MPa}}{0.48} < \frac{0.9 f_u}{\gamma_{M2}} = 259.2 \text{ MPa} \quad \text{OK}$$

$$\sigma_{\perp} = \tau_{\perp} = \frac{M_{y,d} z_3 / (I_w \sqrt{2}) - N_d / (A_w \sqrt{2})}{\text{Utilization}} = \frac{125.5 \text{ MPa}}{0.48} < \frac{0.9 f_u}{\gamma_{M2}} = 259.2 \text{ MPa} \quad \text{OK}$$

$$\sqrt{\sigma_{\perp}^2 + 3(\tau_{\perp}^2 + \tau_{\parallel}^2)} = 264.5 \text{ MPa} < \frac{f_u}{\beta_w \gamma_{M2}} = 360.0 \text{ MPa} \quad \text{OK}$$

Point ④

$$\sigma_{\perp} = \tau_{\perp} = \frac{M_{y,d} z_4 / (I_w \sqrt{2}) - N_d / (A_w \sqrt{2})}{\text{Utilization}} = \frac{139.8 \text{ MPa}}{0.54} < \frac{0.9 f_u}{\gamma_{M2}} = 259.2 \text{ MPa} \quad \text{OK}$$

$$\sqrt{\sigma_{\perp}^2 + 3 \tau_{\perp}^2} = 279.5 \text{ MPa} < \frac{f_u}{\beta_w \gamma_{M2}} = 360.0 \text{ MPa} \quad \text{OK}$$

Weld is OK

Web panel in shear 6.2.6.1 Considered internal forces: in column axis

Slenderness of the panel:
 $d/t_w = 17.7 < 69 \epsilon = 69.0$ **OK** 6.2.6.1(1)

Design shear load:
 $V_{wp,Ed} = M_{y,Ed} / (h - t_{fl,1} / 2 - t_{fl,2} / 2) = 487$ kN

Design web panel resistance:
 $V_{wp,Rd} = 0.9 t_{y,wc} A_{wc} / (\sqrt{3} \gamma_{M0}) = 454$ kN 6.2.6.1(6.7)

Contribution of stiffeners+flanges frame:

Column flanges plastic resistance in bending:
 $M_{pl,fc1,Rd} = 0.25 b_1 t_{fl,1}^2 f_{y,fc} / \gamma_{M0} = 4.7$ kNm
 $M_{pl,fc2,Rd} = 0.25 b_2 t_{fl,2}^2 f_{y,fc} / \gamma_{M0} = 4.7$ kNm

Stiffeners plastic resistance in bending:
 $M_{pl,st,Rd} = 0.25 b_{vh} t_{vh}^2 f_{y,vh} / \gamma_{M0} = 0.88125$ kNm
 $M_{pl,sl,Rd} = 0.25 b_{vd} t_{vd}^2 f_{y,vd} / \gamma_{M0} = 0.88125$ kNm

Contribution of frame behavior to the total shear resistance:
 $V_{wp,add,Rd,1} = 4 M_{pl,fc,Rd} / d_s = 59$ kN Distance between stiffeners:
 $V_{wp,add,Rd,2} = (2 M_{pl,fc,Rd} + 2 M_{pl,st,Rd}) / d_s = 35$ kN $d_s = 319$ mm
 Minimum of both values: $V_{wp,add,Rd,min} = 35$ kN 6.2.6.1(6.8)

Assessment:
 $V_{wp,Rd} + V_{wp,add,Rd,min} = 489$ kN $>$ $V_{wp,Ed} = 487$ kN
 Utilization

1.00	<	1.00
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OK

Column web in compression 6.2.6.2 Considered internal forces: by column flange

Design compression load:
 $N_{c,wc,Ed} = M_{y,Ed,pl,sm} / (h - t_{fl,1} / 2 - t_{fl,2} / 2) - N_{Ed}^* A_{fb} / A_b = 425$ kN

$b_{eff,c,wc} = t_{fb} + 2\sqrt{2} a_b + 5(t_{fc} + s) = 236$ mm 6.2.6.2(6.10)

Reduction factor ω considering the possible effect of interaction with shear:
 $\beta = 1 \Rightarrow \omega = 1 / \sqrt{1 + 1.3 (b_{eff,c,wc} t_{wc} / A_{wc})^2} = 0.695$ 6.2.6.2(tab. 6.3)

Reduction buckling factor 6.2.6.2(6.13):

Plate slenderness: $\lambda_p = 0.932 \sqrt{(b_{eff,c,wc} d_{wc} f_{y,wc} / (E t_{wc}^2))} = 0.637$

Buckling factor: $\rho = (\lambda_p - 0.2) / \lambda_p^2 = 1.000$

Reduction factor k_{wc} 6.2.6.2(2):

Longitudinal compressive stress in column wall: $\sigma_{com,Ed} = M_{y,Ed} / I_c^* d_{wc} / 2 + V_{z,Ed} / A_c = 96$ MPa $<$ $0.7 f_{y,wc} = 164.5$ MPa

$k_{wc} = 1$

Design resistance of unstiffened web in compression:
 $F_{c,wc,Rd} = \rho \omega k_{wc} b_{eff,c,wc} t_{wc} f_{y,wc} / \gamma_{M0} = 385$ kN 6.2.6.2(6.9)

Design resistance of stiffened web in compression:
 $F_{c,wc,Rd} = \omega k_{wc} A_{eff,c,wc} f_{y,wc} / \gamma_{M0} = 630$ kN $A_{eff,c,wc} = b_{eff,c,wc} t_{wc} + b_{vd} t_{vd} = 3860$ mm²

Assessment:
 $F_{c,wc,Rd} = 630$ kN $>$ $N_{c,wc,Ed} = 425$ kN
 Utilization

0.67	<	1.00
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OK

Column web in tension		6.2.6.3 Considered internal forces: by column flange							
Design tension load:									
$N_{t,wc,Ed} = M_{y,Ed,passtr} / ((h - t_{fl1}) / 2 - t_{fl2} / 2) + N_{Ed} * A_{t0} / A_b =$	425	kN							
$b_{eff,t,wc} = t_{fb} + 2 \sqrt{2} b_b + 5(t_{fc} + s) =$	236	mm	6.2.6.3(6.16)						
Reduction factor ω considering the possible effect of interaction with shear:									
$\beta =$	1	=>	$\omega = \omega_1 = 1 / \sqrt{1 + 1,3(b_{eff,t,wc} t_{wc} / A_{vc})^2} =$ 0.695 6.2.6.2(tab. 6.3)						
Design resistance of unstiffened web in tension:									
$F_{t,wc,Rd} = \omega b_{eff,t,wc} t_{wc} f_{y,wc} / \gamma_{M0} =$	385	kN	6.2.6.3(6.15)						
Design resistance of stiffened web in tension:									
$F_{t,wc,Rd} = \omega A_{eff,t,wc} f_{y,wc} / \gamma_{M0} =$	630	kN	$A_{eff,t,wc} = b_{eff,t,wc} t_{wc} + b_{vh} t_{vh} =$ 3860 mm ²						
Assessment:									
$F_{t,wc,Rd} =$	630	kN	>						
Utilization	<table border="1"> <tr> <td>$N_{t,wc,Ed} =$</td> <td>425</td> <td>kN</td> </tr> <tr> <td>0.67</td> <td><</td> <td>1.00</td> </tr> </table>		$N_{t,wc,Ed} =$	425	kN	0.67	<	1.00	OK
$N_{t,wc,Ed} =$	425	kN							
0.67	<	1.00							
Column flange in bending		6.2.6.4 Considered internal forces: by column flange							
By upper beam flange - tension zone:									
Design load: $N_{fc,Ed} = N_{t,wc,Ed} =$ 0 kN The flange is stiffened, is not examined!!									
$b_{eff,b,fc} = b_{1,eff} =$	160	mm	4.10 (2)						
Design resistance of unstiffened flange in bending: 6.2.6.4.3									
$F_{fc,Rd} = b_{eff,b,fc} t_{fb} f_{y,fb} / \gamma_{M0} =$	432	kN							
Assessment:									
$F_{fc,Rd} =$	432	kN	>						
Utilization	<table border="1"> <tr> <td>$N_{fc,Ed} =$</td> <td>0</td> <td>kN</td> </tr> <tr> <td>0.00</td> <td><</td> <td>1.00</td> </tr> </table>		$N_{fc,Ed} =$	0	kN	0.00	<	1.00	OK
$N_{fc,Ed} =$	0	kN							
0.00	<	1.00							
By lower beam flange - compression zone:									
Design load: $N_{fc,Ed} = N_{c,wc,Ed} =$ 0 kN The flange is stiffened, is not examined!!									
$b_{eff,b,fc} = b_{2,eff} =$	160	mm	4.10 (2)						
Design resistance of unstiffened flange in bending: 6.2.6.4.3									
$F_{fc,Rd} = b_{eff,b,fc} t_{fb} f_{y,fb} / \gamma_{M0} =$	432	kN							
Assessment:									
$F_{fc,Rd} =$	432	kN	>						
Utilization	<table border="1"> <tr> <td>$N_{fc,Ed} =$</td> <td>0</td> <td>kN</td> </tr> <tr> <td>0.00</td> <td><</td> <td>1.00</td> </tr> </table>		$N_{fc,Ed} =$	0	kN	0.00	<	1.00	OK
$N_{fc,Ed} =$	0	kN							
0.00	<	1.00							
Beam flange in compression		6.2.6.7 Considered internal forces: by column flange							
Design load: $N_{c,fb,Ed} = N_{c,wc,Ed} =$ 425 kN									
D. res. of beam in bending: $M_{c,Rd} =$ 189 kNm Class of beam: 1									
Design resistance: $F_{c,fb,Rd} = M_{c,Rd} / (h - t_{fl1} / 2 - t_{fl2} / 2) =$ 593 kN 6.2.6.7 (6.21)									
Assessment:									
$F_{c,fb,Rd} =$	593	kN	>						
Utilization	<table border="1"> <tr> <td>$N_{c,fb,Ed} =$</td> <td>425</td> <td>kN</td> </tr> <tr> <td>0.72</td> <td><</td> <td>1.00</td> </tr> </table>		$N_{c,fb,Ed} =$	425	kN	0.72	<	1.00	OK
$N_{c,fb,Ed} =$	425	kN							
0.72	<	1.00							

4 Results by CBFEM Idea RS software

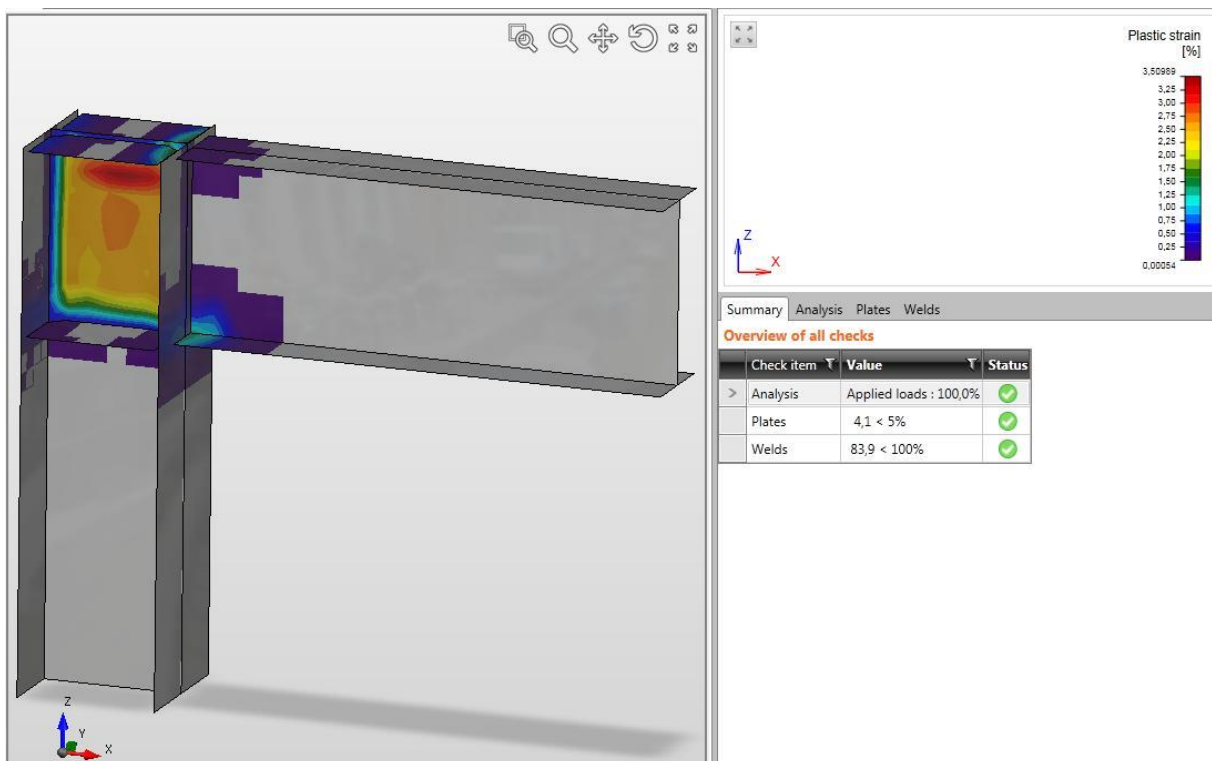
CBFEM - combination of the advantages of finite element method and analytical component method.

Shell elements, special spring and contact elements with characteristics according to the component method.

Elastic-plastic stress-strain diagram for material of shell elements. Assessment is based on the maximum strain given according to EN1993-1-5 by value of 5%.

Bolts are modelled using special spring elements and assessment is carried out according to standard procedures described in EN1993-1-8.

Result of Idea RS software for the beam IPE330 to column HEB260 connection is shown below.



CON1

Summary

Name	Value	Check status
Analysis	Applied loads : 100,0%	OK
Plates	4,1 < 5%	OK
Welds	83,9 < 100%	OK

Plates

Name	Thickness [mm]	Load case	σ_{Ed} [MPa]	ϵ_{Pl} [1e-4]	Check status
C-bfl	18	LE1	237,28	108,4	OK
C-tfl	18	LE1	237,54	120,6	OK
C-web	10	LE1	243,64	411,2	OK
B-bfl	12	LE1	237,77	131,6	OK
B-tfl	12	LE1	235,20	9,5	OK
B-web	8	LE1	237,57	122,5	OK
STIFF1	10	LE1	235,48	23,0	OK
STIFF1	10	LE1	235,48	23,0	OK
STIFF1	10	LE1	238,42	162,6	OK
STIFF1	10	LE1	238,42	162,6	OK

Design data

	f_y [MPa]	ϵ_{lim} [1e-4]
S 235	235,00	500,0

Symbol explanation

Symbol	Symbol explanation
ϵ_{Pl}	Strain
σ_{Ed}	Eq. stress

Welds

Name	Edge	Size [mm]	Load case	$\sigma_{w,Ed}$ [MPa]	σ_{\perp} [MPa]	Ut [%]	Check status
C-tfl	B-bfl	12	LE1	141,12	-91,94	39,20	OK
C-tfl	B-tfl	12	LE1	151,85	72,34	42,18	OK
C-tfl	B-web	8	LE1	54,86	-22,23	15,24	OK
C-bfl	STIFF1	8	LE1	127,18	-88,25	35,33	OK
C-web	STIFF1	8	LE1	134,11	-13,74	37,25	OK
C-tfl	STIFF1	8	LE1	174,06	60,04	48,35	OK
C-bfl	STIFF1	8	LE1	161,94	-52,56	44,98	OK
C-web	STIFF1	8	LE1	63,49	40,85	17,64	OK
C-tfl	STIFF1	8	LE1	141,03	93,85	39,18	OK
C-bfl	STIFF1	8	LE1	94,10	-22,27	26,14	OK
C-web	STIFF1	8	LE1	245,62	65,13	68,23	OK
C-tfl	STIFF1	8	LE1	269,20	-158,47	74,78	OK
C-bfl	STIFF1	8	LE1	57,13	21,43	15,87	OK
C-web	STIFF1	8	LE1	259,30	-28,09	72,03	OK
C-tfl	STIFF1	8	LE1	301,99	-125,50	83,88	OK

Design data

	β_w	$\sigma_{w,Rd}$ [MPa]	$0,9 \sigma_{w,Rd}$ [MPa]
S 235		360,00	259,20

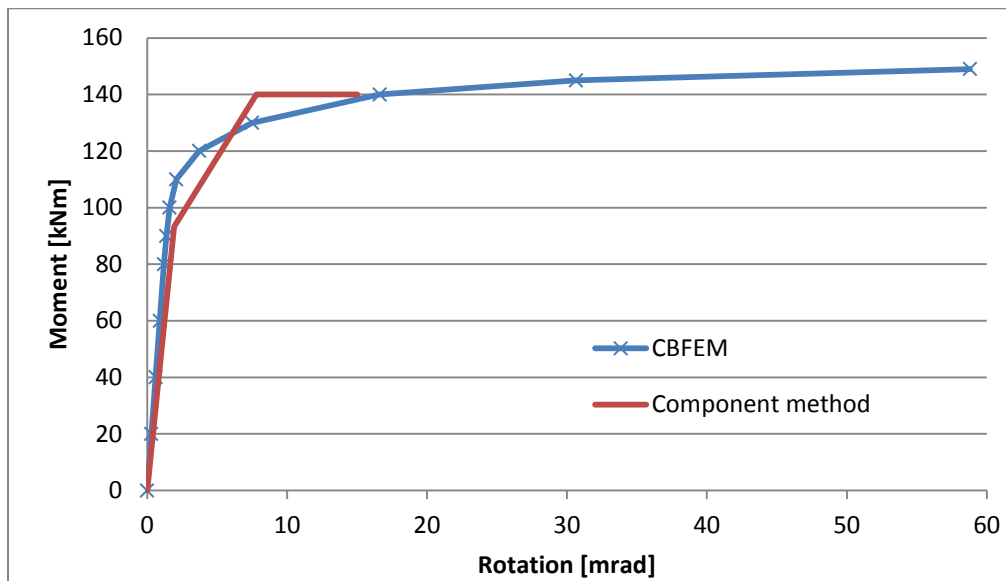
Symbol explanation

Symbol	Symbol explanation
$\sigma_{w,Ed}$	Equivalent stress
$\sigma_{w,Rd}$	Equivalent stress resistance
σ_{\perp}	Perpendicular stress
$0,9 \sigma_{w,Rd}$	Perpendicular stress resistance
β_w	Corelation factor EN 1993-1-8 tab. 4.1
Ut	Utilization

5 Global behaviour and verification

Comparison of the global behaviour of the joint described by moment-rotation diagrams for both design procedures mentioned above was done. Attention was focused on the main characteristics of the moment-rotation diagram: initial stiffness, elastic resistance and design resistance. Connection of the beam IPE330 to column HEB260 was chosen as a sample. Joint with horizontal stiffeners in column is considered according to component method as a rigid joint with $S_{j,ini} = \infty$. For this reason, a joint without horizontal stiffeners in column is used for this global behaviour study. Results of both design procedures are shown in the graph and the table below. Both procedures give for initial

stiffness, elastic resistance and design resistance similar results. In addition, maximal rotation is compared. Component method gives for maximal rotation only guaranteed minimal value 15 mrad see EN 1993-1-8 - 6.4.3(2) and that is the reason of much lower value compared to CBFEM.



		CM	CBFEM	CM/CBFEM
Initial stiffness	[kNm/rad]	48423.7	66889.6	0.72
Elastic resistance	[kNm]	93.3	90.0	1.04
Design resistance	[kNm]	140.0	149.0	0.94
Maximal rotation	[mrad]	15.0	58.8	0.26

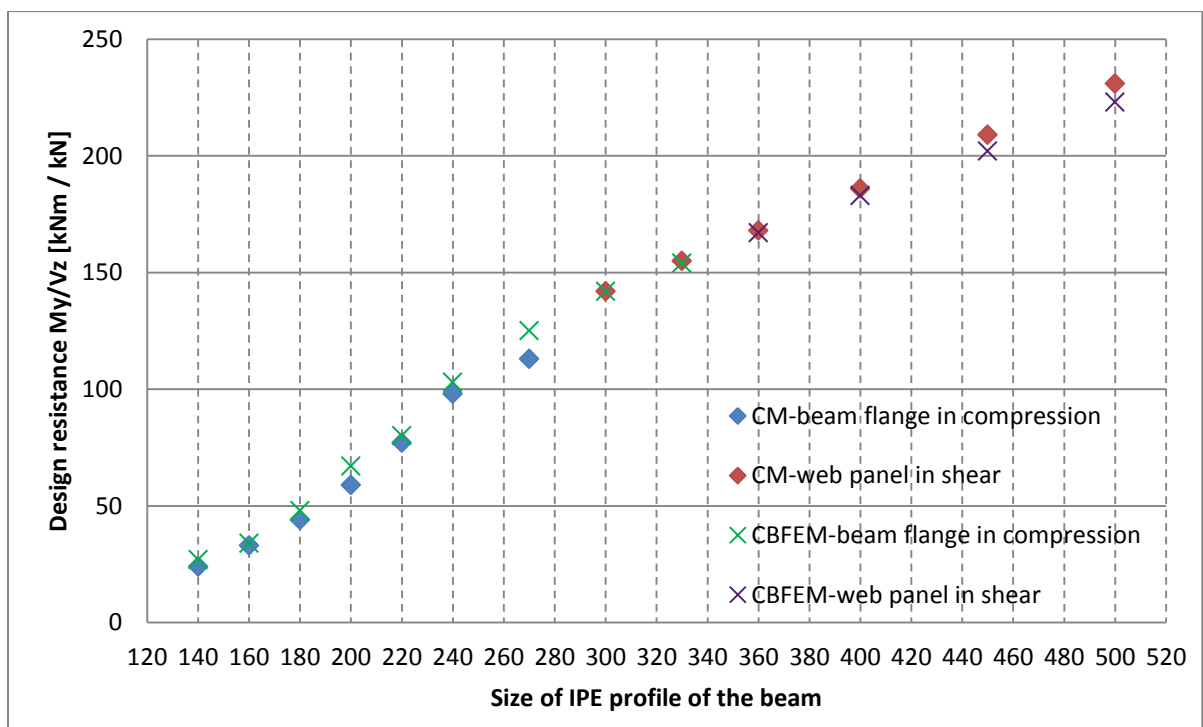
6 Verification of resistance

Design resistance calculated by CBFEM Idea RS software were compared with the results of the component method in the next step. The comparison was focused on capacity and also to determine the critical component.

The study was performed for three different parameters: beam cross-section, column cross section and thickness of the column wall.

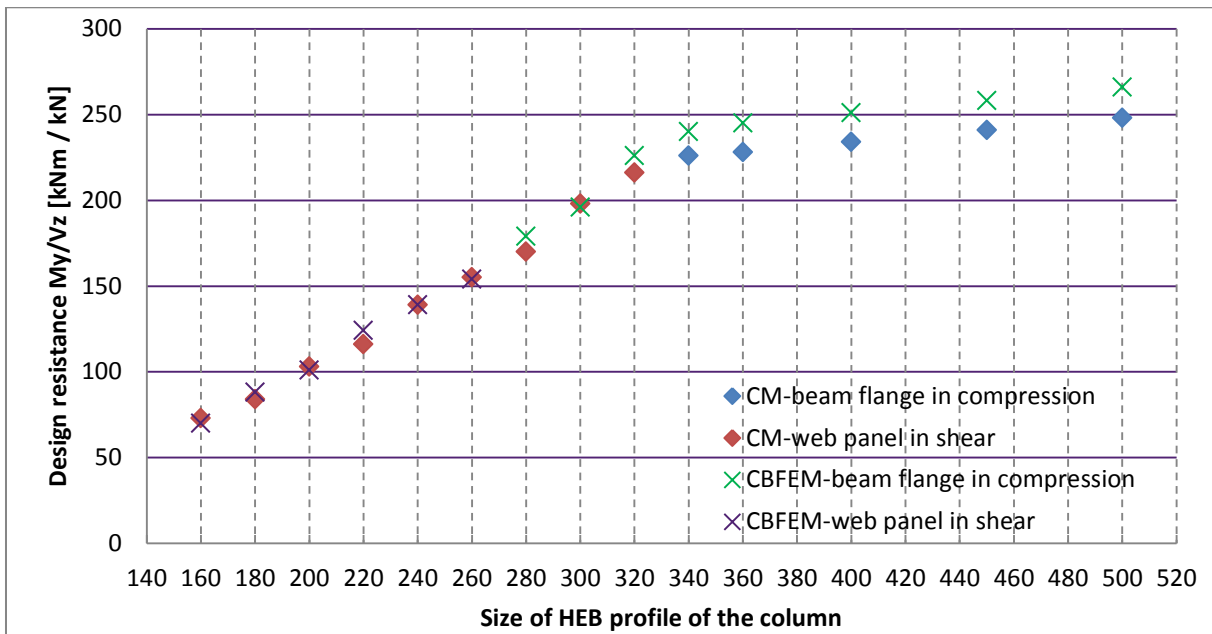
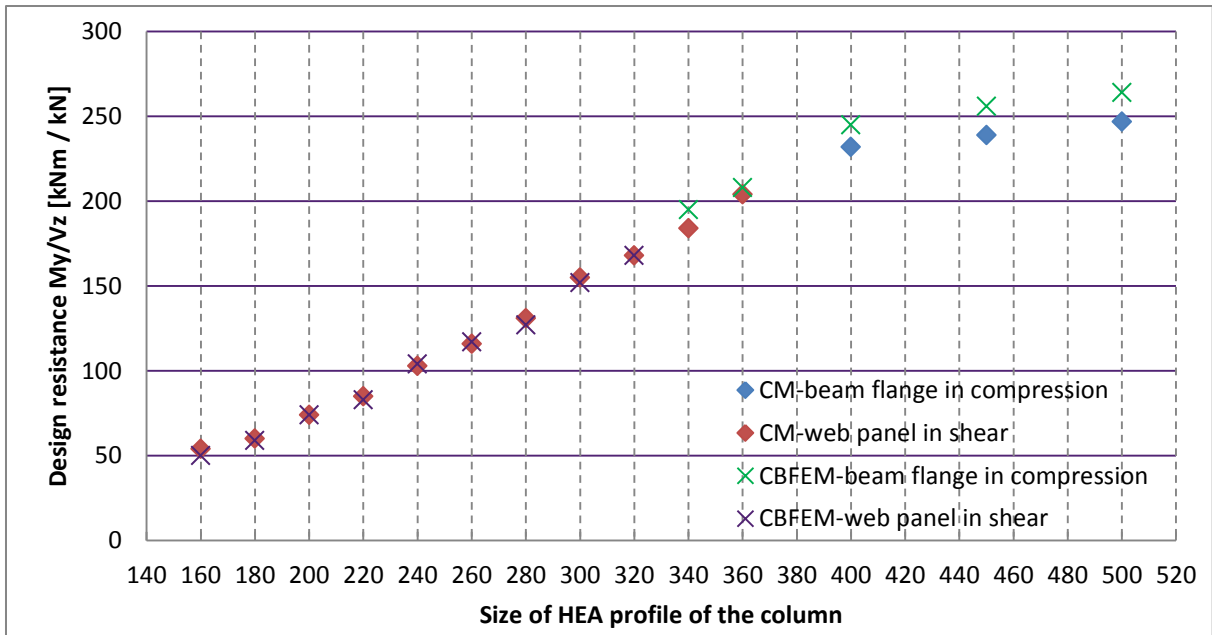
In the first case with parameter beam cross-section was a column cross-section HEB260 and horizontal column stiffener thick was 10 mm and width corresponding to the width of beam flange. IPE sections were selected for horizontal beam from IPE140 to IPE500. The results are shown in table and graph.

Parameter	Component method		CBFEM-Idea RS	
	Resistance [kN/kNm]	Critical component	Resistance [kN/kNm]	Critical component
IPE140	24	Beam flange in compression	27	Beam flange in compression
IPE160	33	Beam flange in compression	34	Beam flange in compression
IPE180	44	Beam flange in compression	48	Beam flange in compression
IPE200	59	Beam flange in compression	67	Beam flange in compression
IPE220	77	Beam flange in compression	80	Beam flange in compression
IPE240	98	Beam flange in compression	103	Beam flange in compression
IPE270	113	Beam flange in compression	125	Beam flange in compression
IPE300	142	Web panel in shear	142	Beam flange in compression
IPE330	155	Web panel in shear	154	Beam flange in compression
IPE360	168	Web panel in shear	167	Web panel in shear
IPE400	186	Web panel in shear	183	Web panel in shear
IPE450	209	Web panel in shear	202	Web panel in shear
IPE500	231	Web panel in shear	223	Web panel in shear



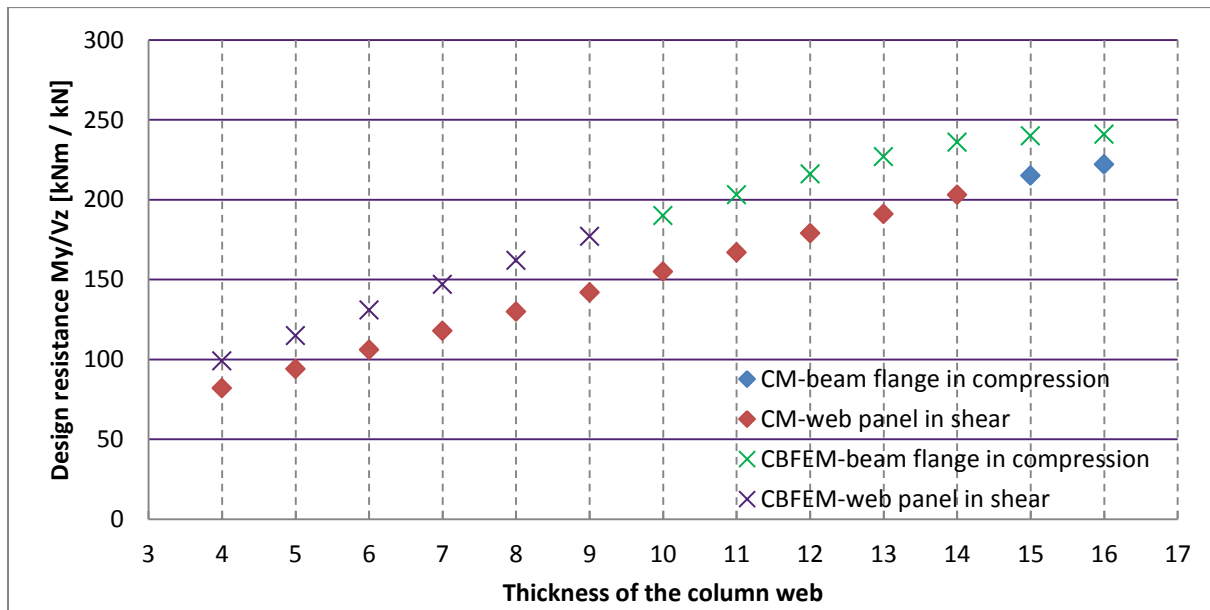
In the second case with parameter column cross-section was a beam cross-section IPE 330 and horizontal column stiffener thick was 10 mm and width 160 mm. HEA and HEB cross-sections were selected for horizontal beam from HEA 160 to HEA 500 and from HEB 160 to HEB 500. The results are shown in table and graphs.

Parameter	Component method		CBFEM-Idea RS	
	Resistance [kN/kNm]	Critical component	Resistance [kN/kNm]	Critical component
HEA160	54	Web panel in shear	50	Web panel in shear
HEA180	60	Web panel in shear	59	Web panel in shear
HEA200	74	Web panel in shear	74	Web panel in shear
HEA220	85	Web panel in shear	83	Web panel in shear
HEA240	103	Web panel in shear	104	Web panel in shear
HEA260	116	Web panel in shear	117	Web panel in shear
HEA280	131	Web panel in shear	127	Web panel in shear
HEA300	155	Web panel in shear	152	Web panel in shear
HEA320	168	Web panel in shear	168	Web panel in shear
HEA340	184	Web panel in shear	195	Beam flange in compression
HEA360	204	Web panel in shear	208	Beam flange in compression
HEA400	232	Beam flange in compression	245	Beam flange in compression
HEA450	239	Beam flange in compression	256	Beam flange in compression
HEA500	247	Beam flange in compression	264	Beam flange in compression
HEB160	73	Web panel in shear	70	Web panel in shear
HEB180	84	Web panel in shear	88	Web panel in shear
HEB200	103	Web panel in shear	101	Web panel in shear
HEB220	116	Web panel in shear	124	Web panel in shear
HEB240	139	Web panel in shear	139	Web panel in shear
HEB260	155	Web panel in shear	154	Web panel in shear
HEB280	170	Web panel in shear	179	Beam flange in compression
HEB300	198	Web panel in shear	196	Beam flange in compression
HEB320	216	Web panel in shear	226	Beam flange in compression
HEB340	226	Beam flange in compression	240	Beam flange in compression
HEB360	228	Beam flange in compression	245	Beam flange in compression
HEB400	234	Beam flange in compression	251	Beam flange in compression
HEB450	241	Beam flange in compression	258	Beam flange in compression
HEB500	248	Beam flange in compression	266	Beam flange in compression

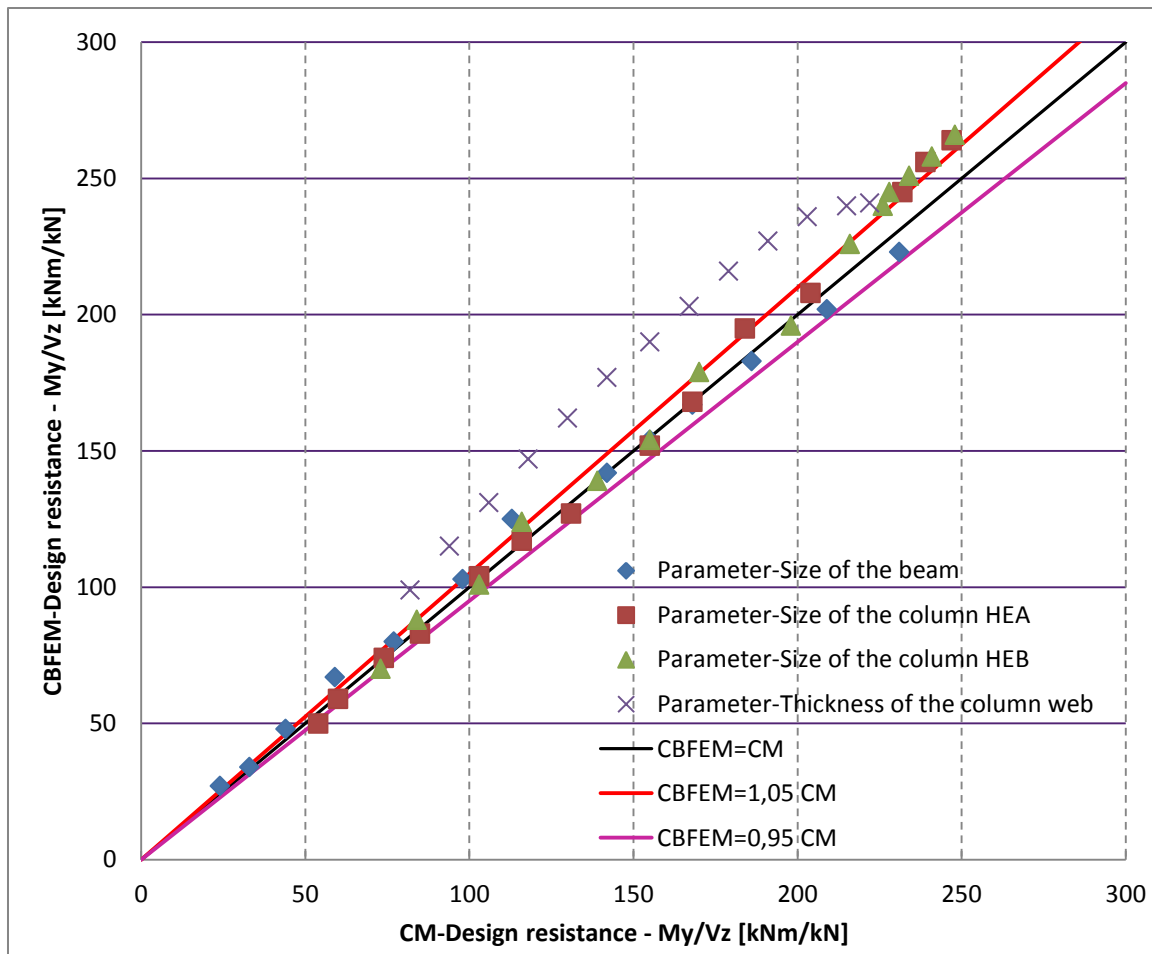


In the third case with parameter thickness of the column wall was a beam cross-section IPE 330 and column cross-section dimensions corresponded to HEA 320 except the wall thickness. Horizontal column stiffener thick was 10 mm and width 160 mm. The wall thickness was chosen from 4 mm to 16 mm with increments of 1 mm. The results are shown in table and graph.

Parameter	Component method		CBFEM-Idea RS	
	Resistance [kN/kNm]	Critical component	Resistance [kN/kNm]	Resistance [kN/kNm]
4	82	Web panel in shear	99	Web panel in shear
5	94	Web panel in shear	115	Web panel in shear
6	106	Web panel in shear	131	Web panel in shear
7	118	Web panel in shear	147	Web panel in shear
8	130	Web panel in shear	162	Web panel in shear
9	142	Web panel in shear	177	Web panel in shear
10	155	Web panel in shear	190	Beam flange in compression
11	167	Web panel in shear	203	Beam flange in compression
12	179	Web panel in shear	216	Beam flange in compression
13	191	Web panel in shear	227	Beam flange in compression
14	203	Web panel in shear	236	Beam flange in compression
15	215	Beam flange in compression	240	Beam flange in compression
16	222	Beam flange in compression	241	Beam flange in compression



To illustrate the accuracy of the CBFEM model, results of the parametric studies were summarized in graph comparing resistance by CBFEM and component method. The results show that the difference of the two calculation methods is up to 5%, which is a generally acceptable value. Except the study with wall thickness parameter where CBFEM model gives higher resistance compared to component method. This difference is caused by considering welded cross-sections. For welded cross-section is in component method for transfer of shear load considered only web and contribution of the flanges is neglected.



7 Reliability

Reliability of CBFEM software is provided in accordance with the strategy of EC considering partial safety factors. Inputs of internal loads are entered as design values with the load factor and combination coefficient. Material safety factors according to EN1993-1-8 are used for design resistance of the connection. For bolts and welds $\gamma_M=1,25$ and for plates $\gamma_M=1,0$.

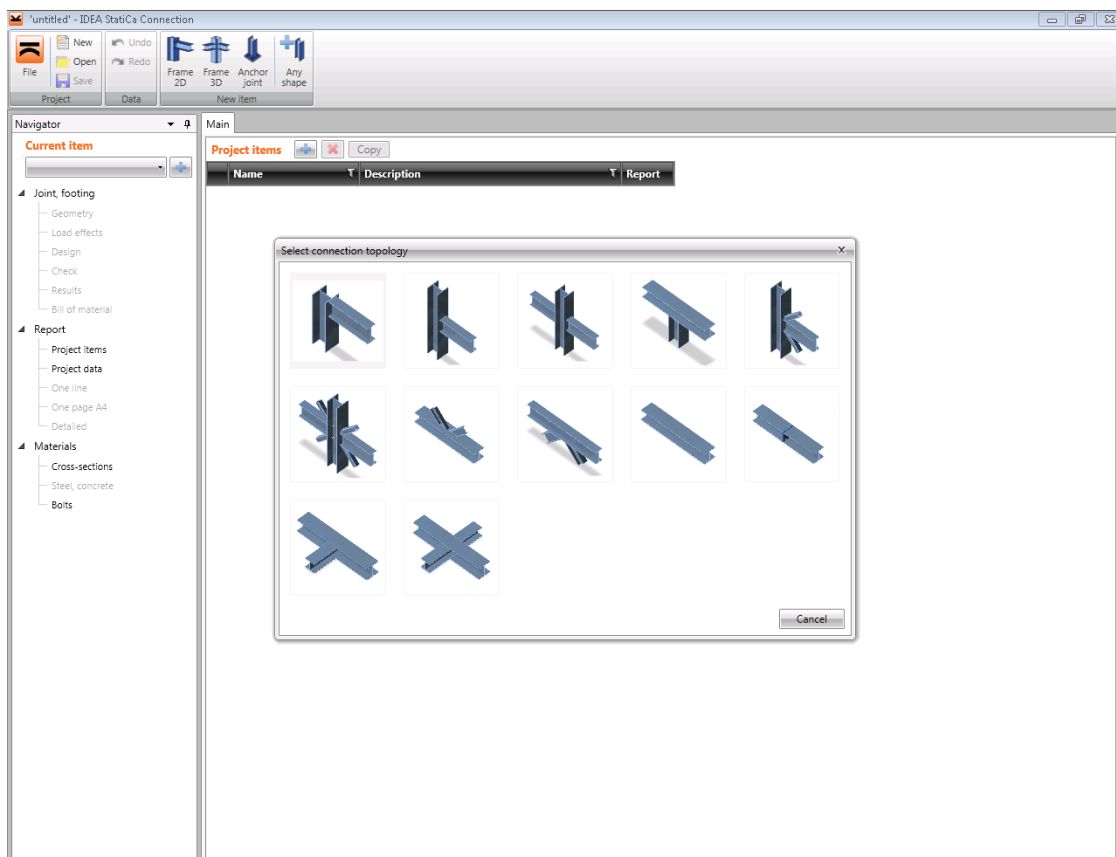
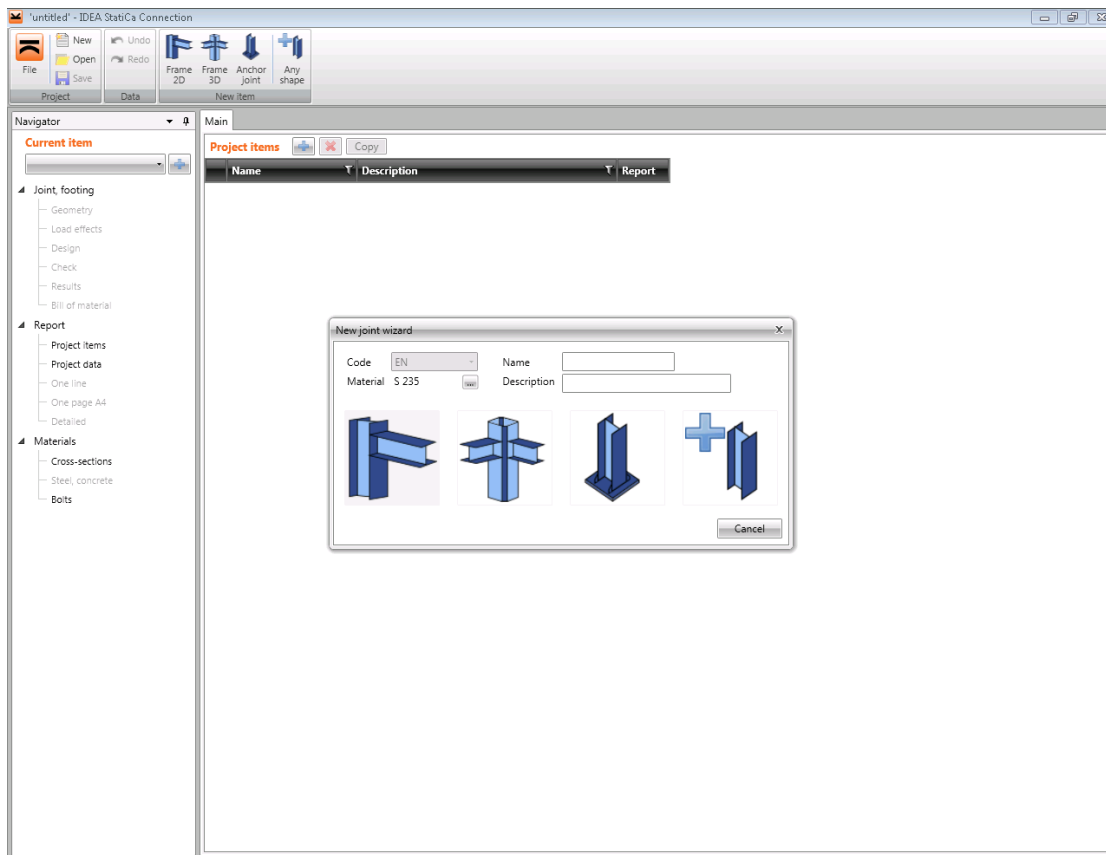
8 Résumé

Verification studies confirmed the accuracy of the CBFEM IDEA RS software. Results of this software were compared with the results of the component method recommended in EN1993-1-8. Both procedures predict similar global behaviour of the joint. Except the study with web thickness parameter is the difference in design resistance of the two calculation methods up to 5%, which is a generally acceptable value. Higher resistance by CBFEM IDEA RS software compared to component method for study with web thickness parameter is caused by conservative calculation of component method for welded cross-sections.

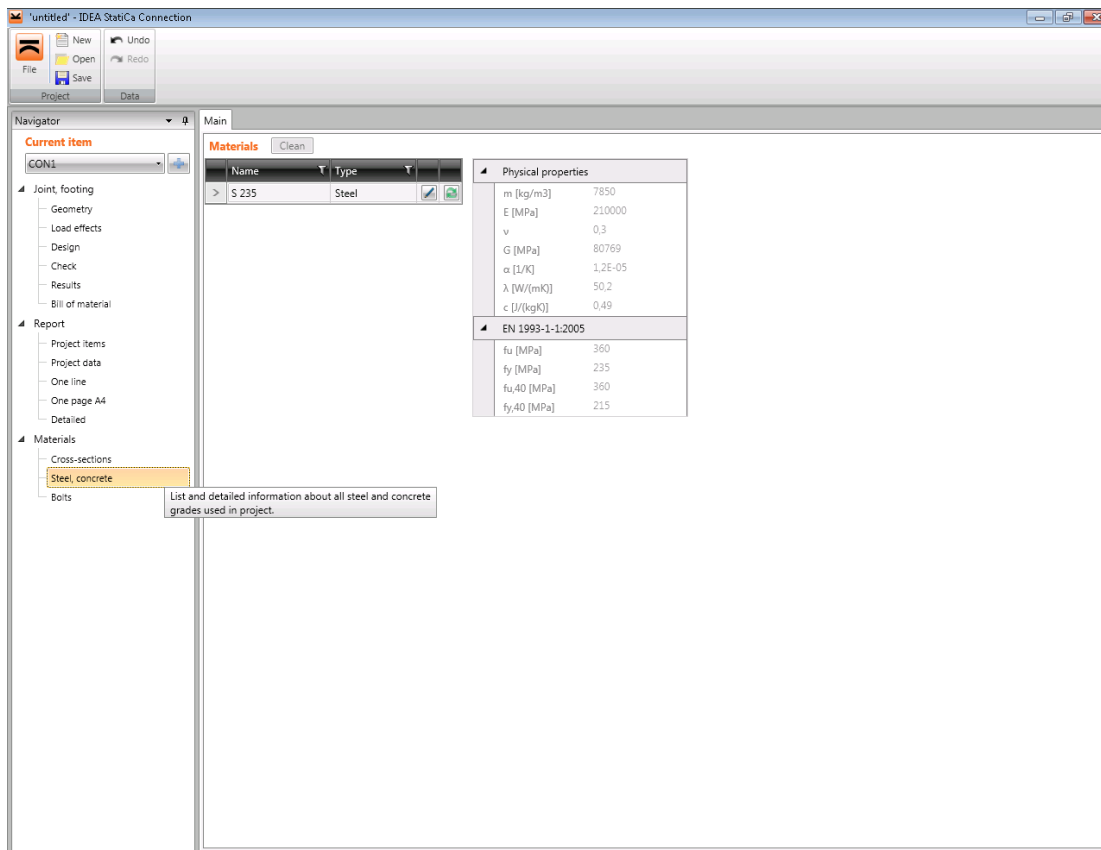
Reliability of CBFEM software is provided in accordance with the strategy of EC considering partial safety factors.

9 Benchmark example

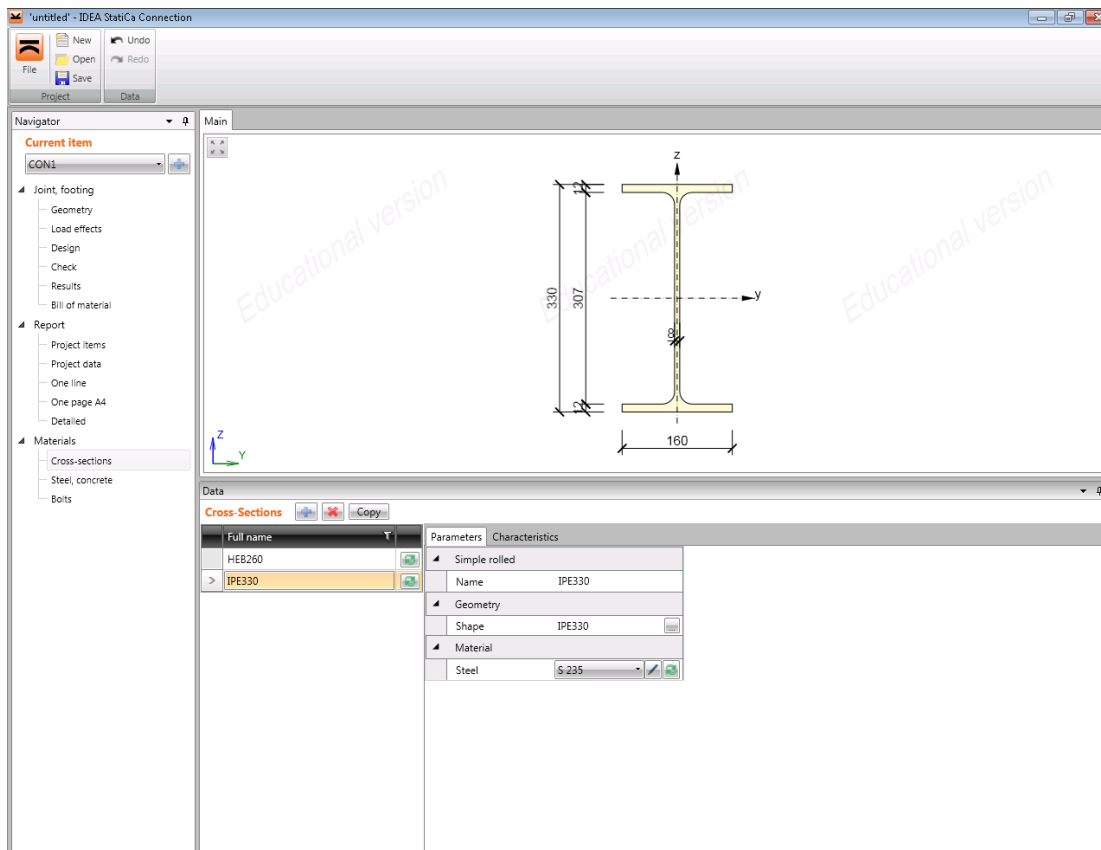
Choice of the geometry:

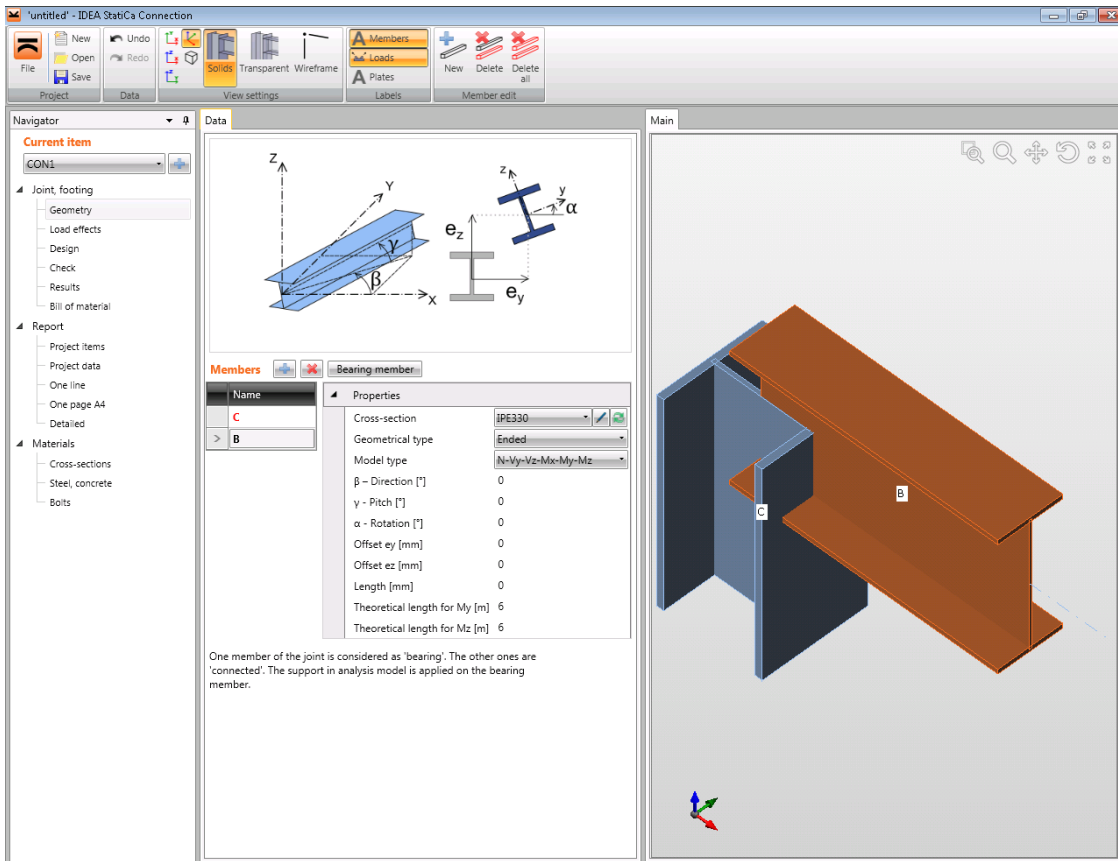


Setting of the material:

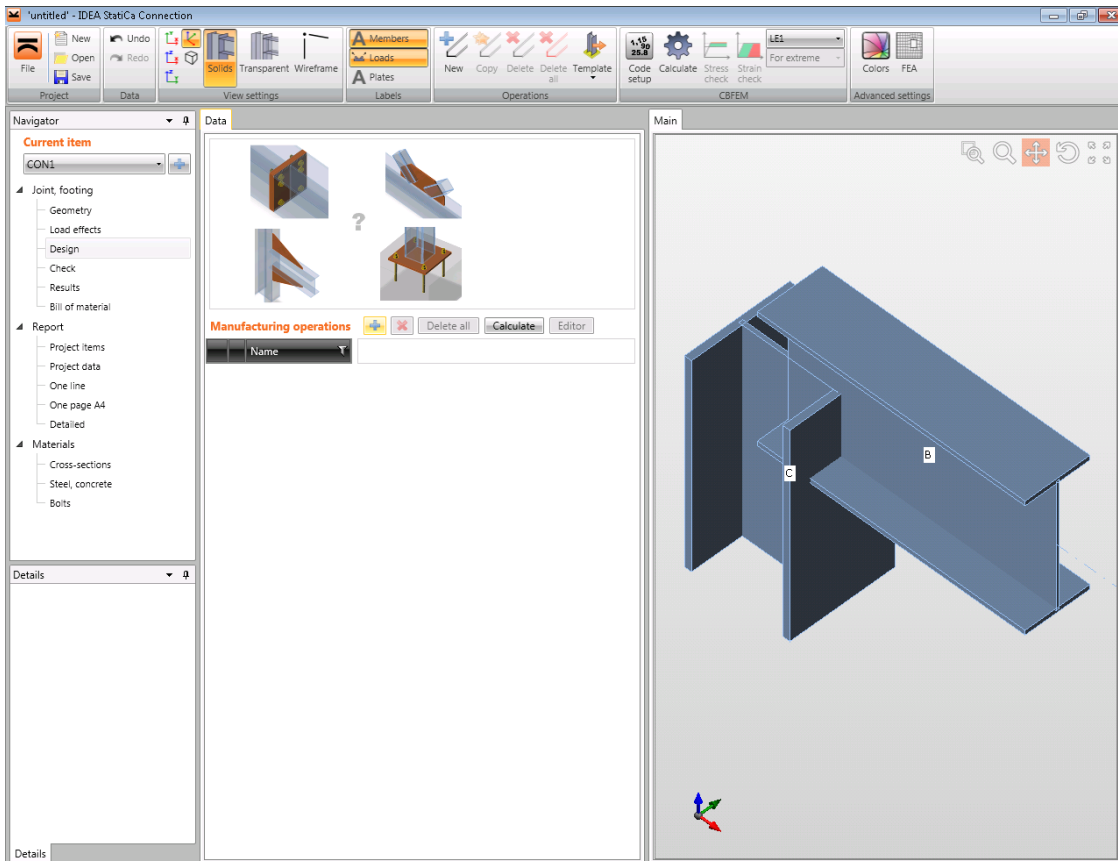


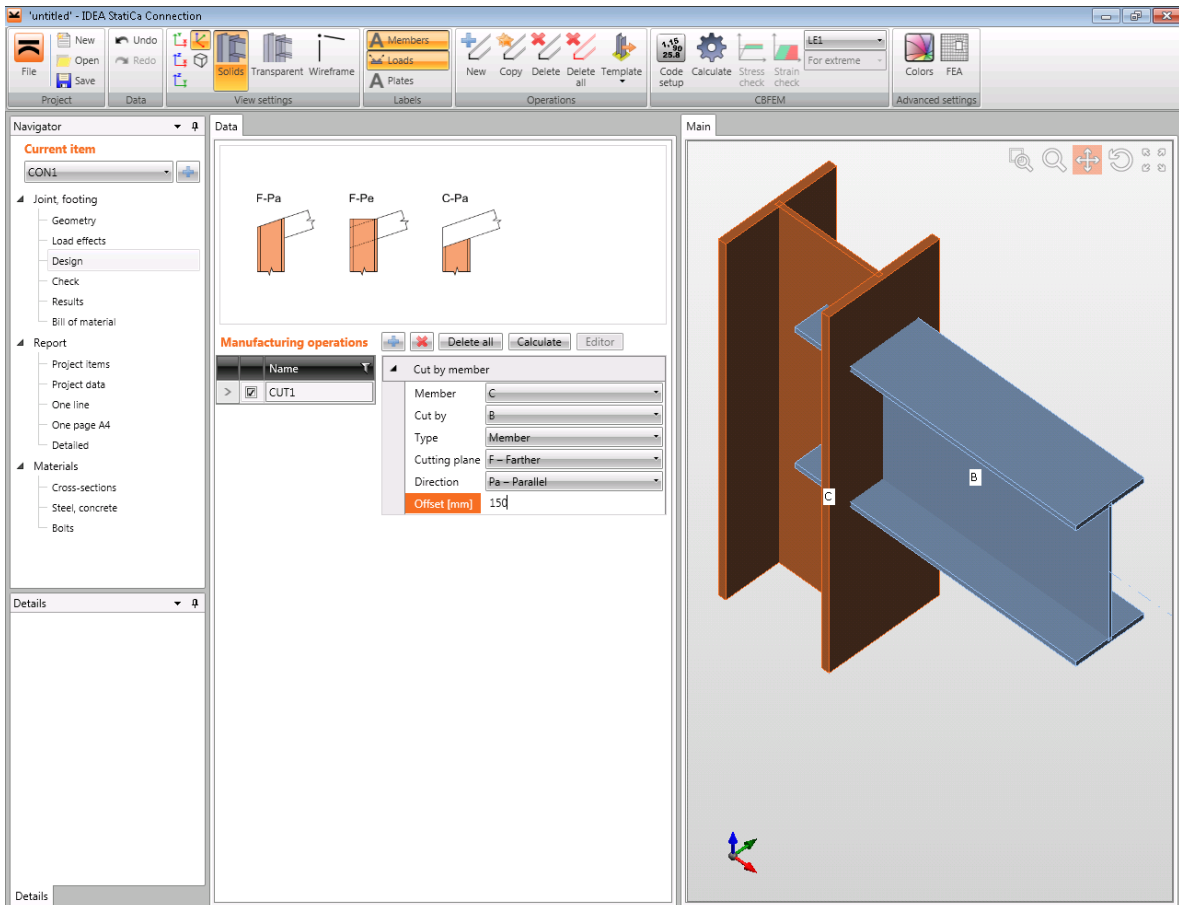
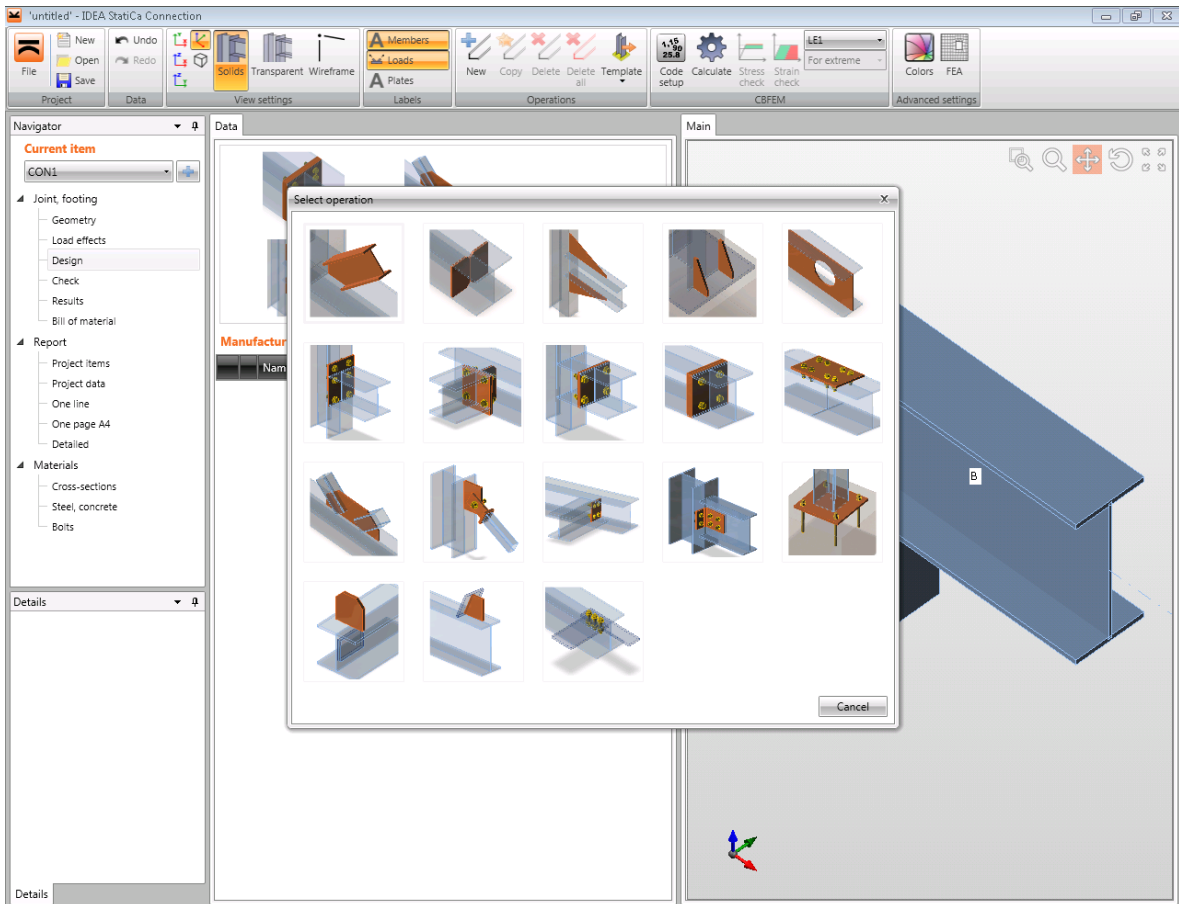
Setting of the cross-sections:

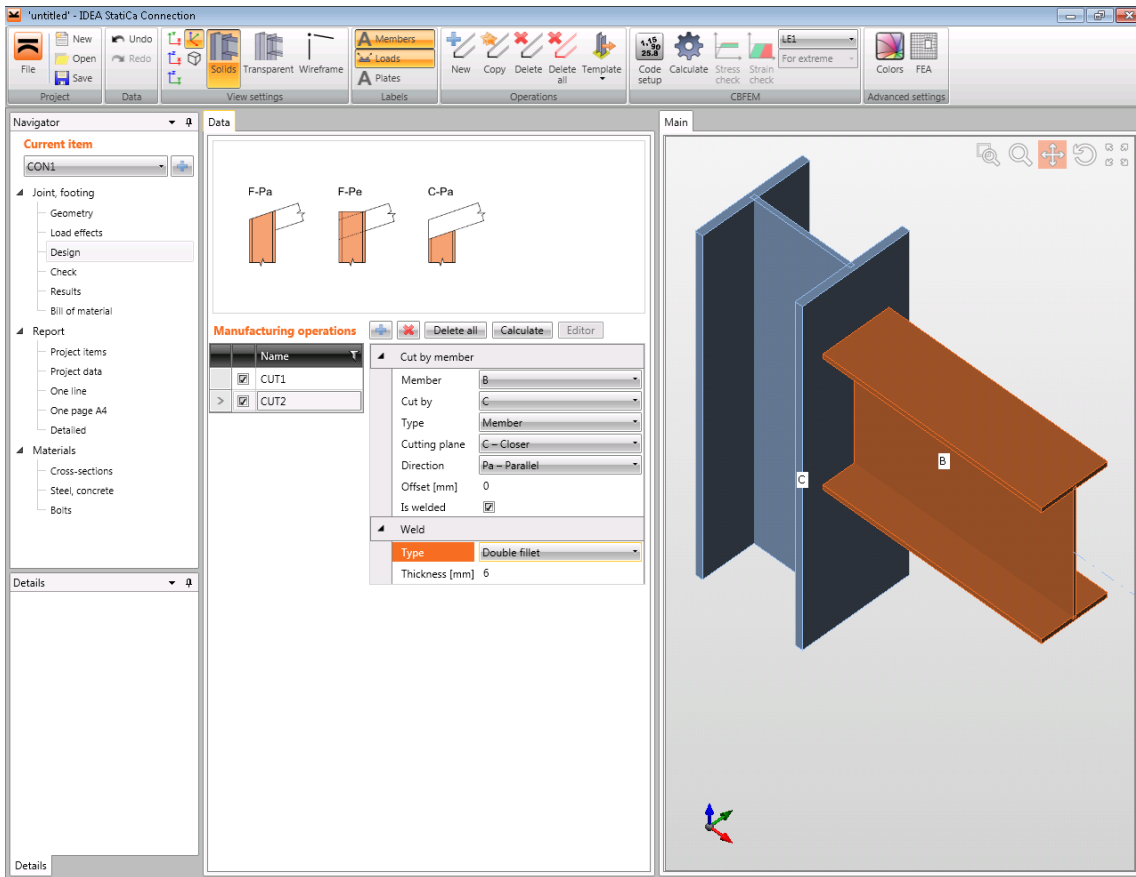




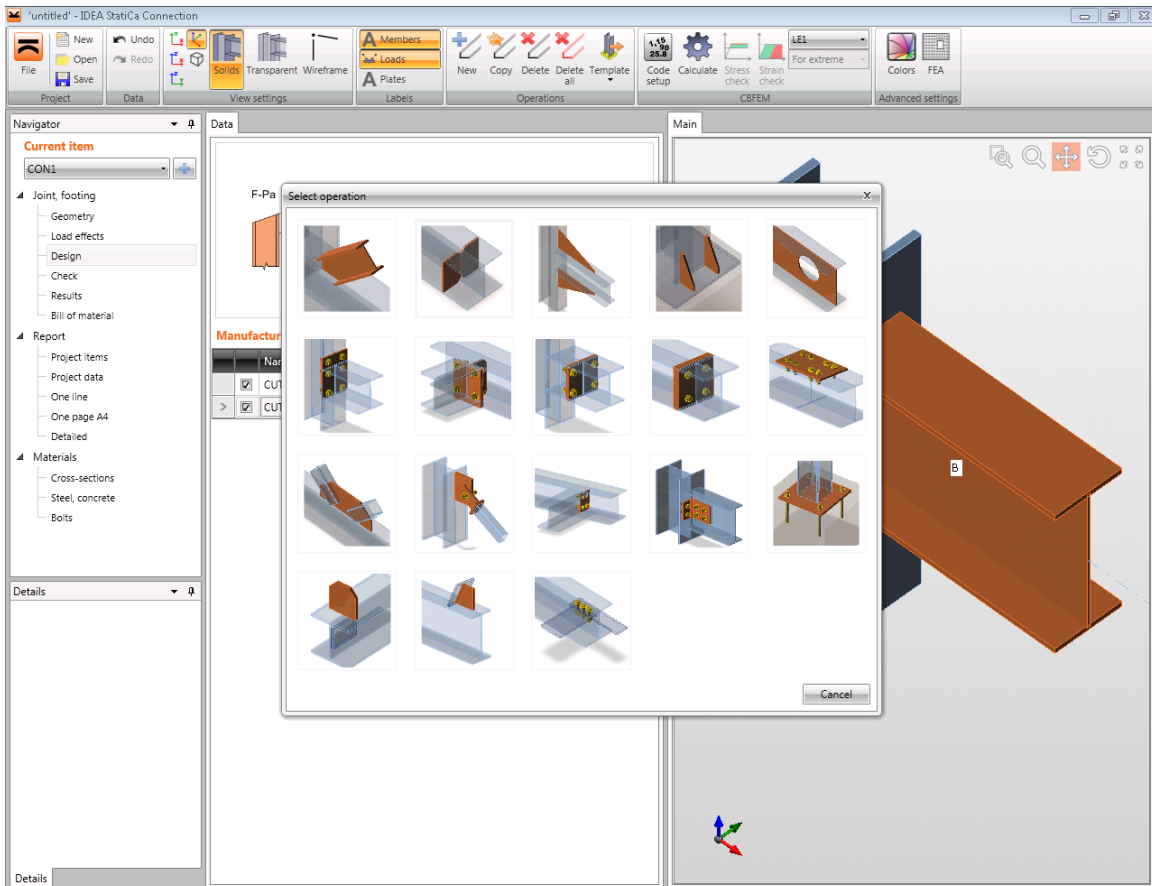
Cutting of the profiles, setting of the welds:

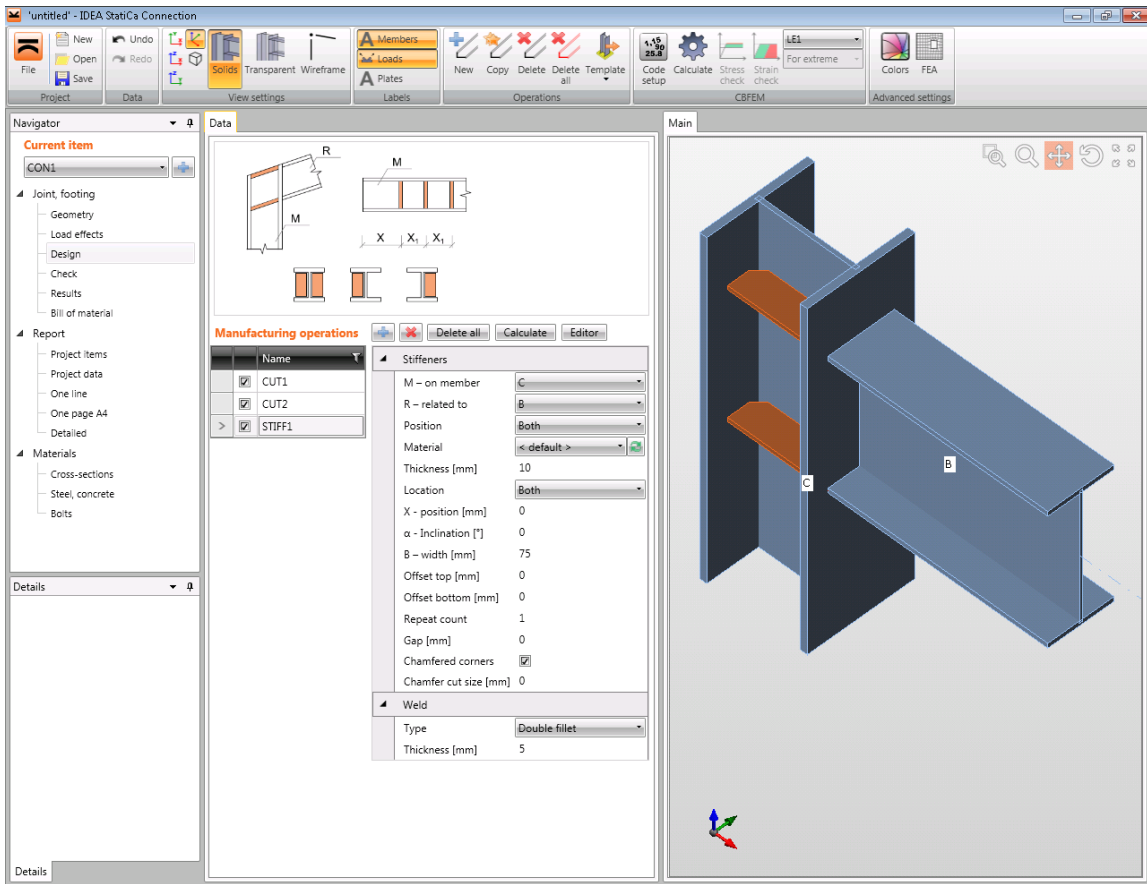




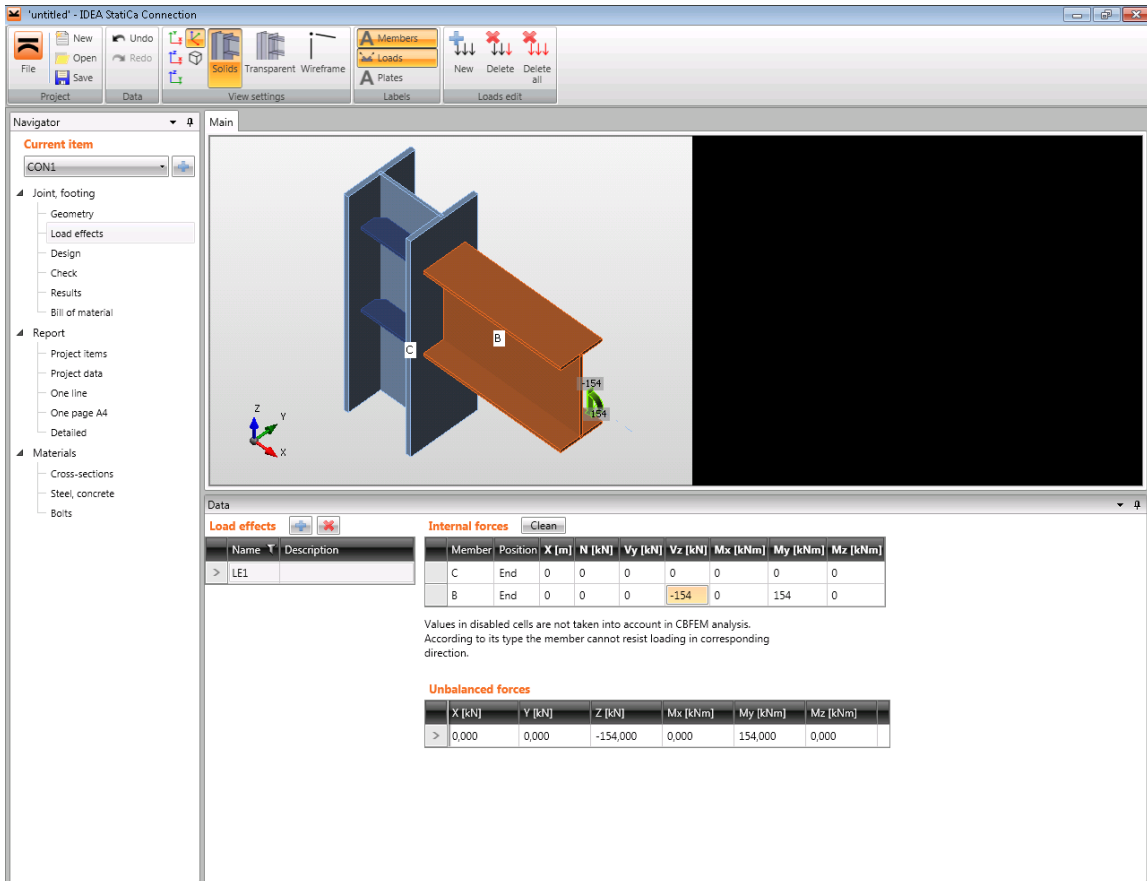


Adding of the horizontal stiffeners of column:

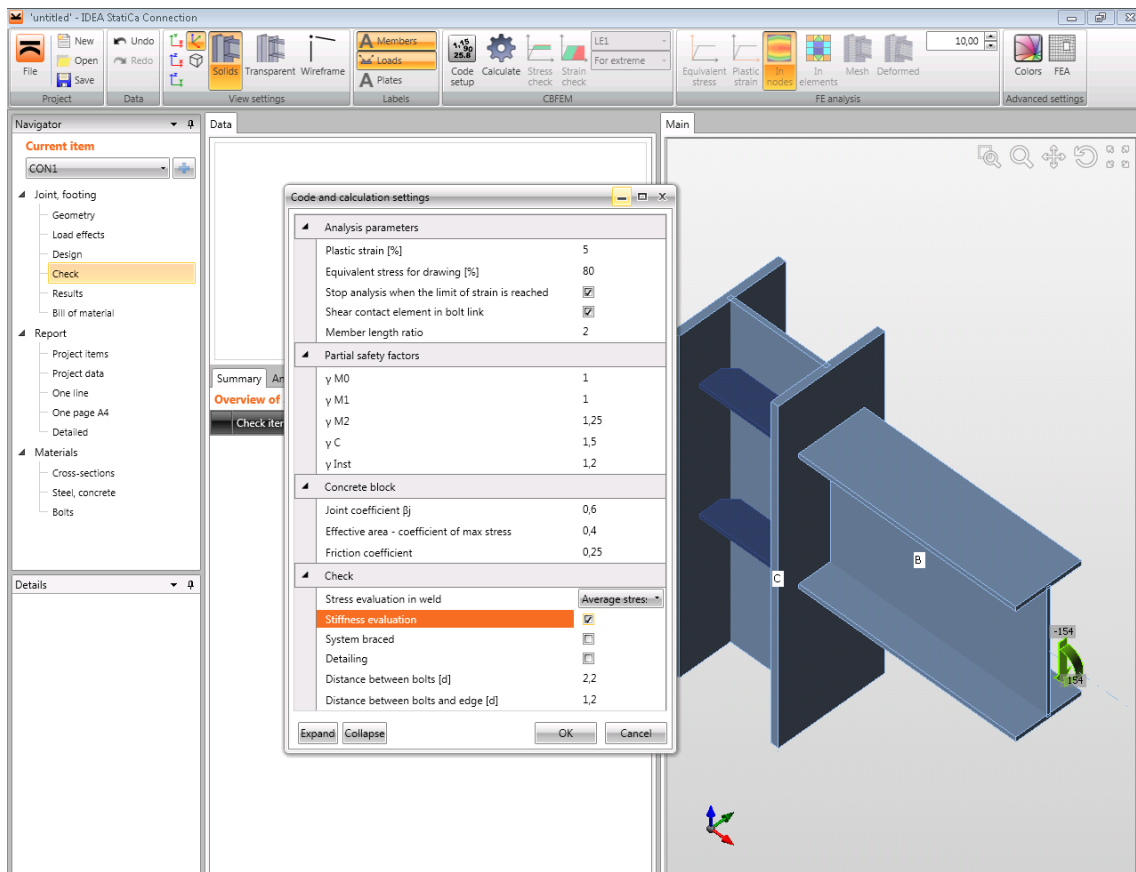
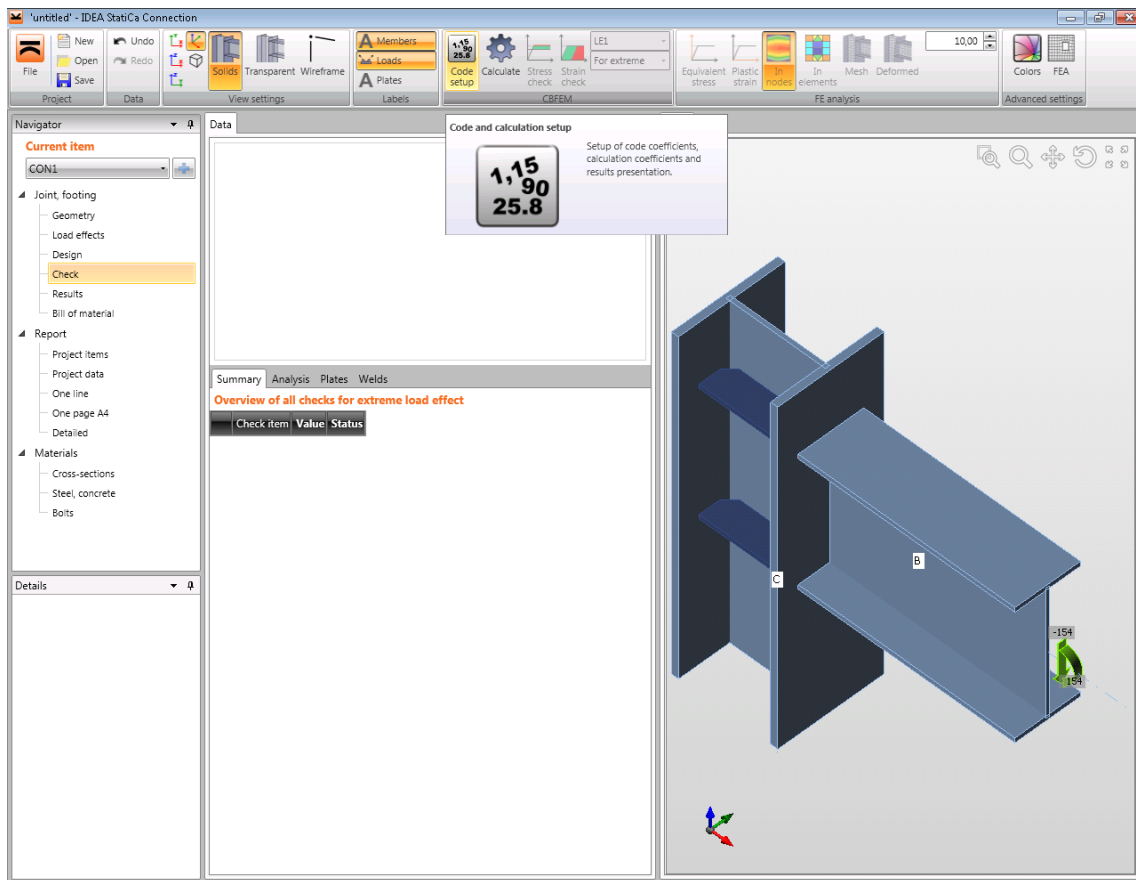


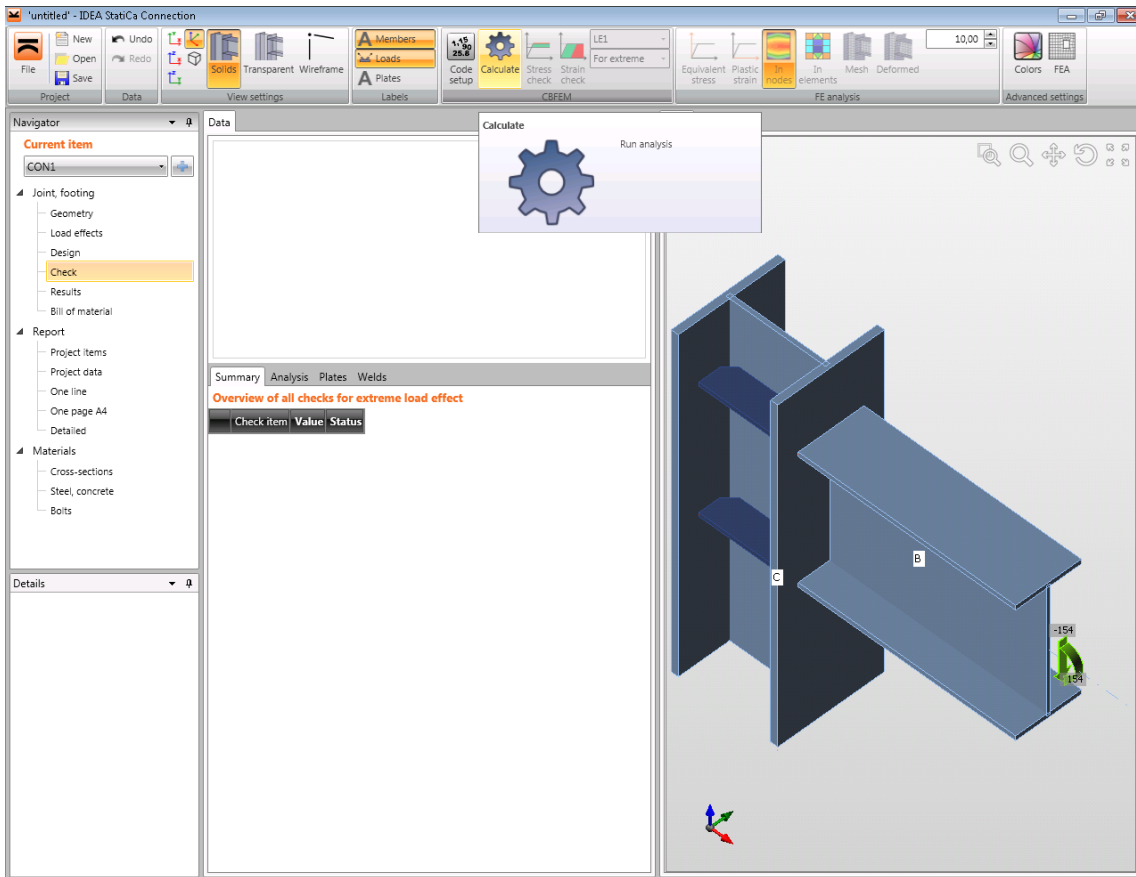


Setting of the loads:



Setting of the calculation:





Results:

The screenshot shows the Results section of the IDEA StatiCa Connection software. The main window displays the 'Results Summary' table, which provides an overview of the analysis results. Below this, there are detailed tables for 'Design data' and 'Welds'. The 'Design data' table shows the yield strength (f_y) and ultimate tensile strength (f_{tm}) for the steel grade S 235. The 'Welds' table provides detailed information for each weld, including its name, edge, size, load case, and various stress and strain values.

Name	Value	Check status
Analysis	Applied loads : 100,0%	OK
Plates	2,3 < 5%	OK
Welds	93,1 < 100%	OK

Name	Thickness [mm]	Load case	σ_{Ed} [MPa]	ϵ_{p1} [1e-4]	Check status
C-flt 1	18	LE1	237,0	93,4	OK
C-flt 1	18	LE1	237,5	117,2	OK
C-web 1	10	LE1	239,8	230,6	OK
B-flt 1	12	LE1	237,5	118,2	OK
B-flt 1	12	LE1	237,4	114,5	OK
B-web 1	8	LE1	237,4	114,3	OK
STIFF1	10	LE1	235,7	35,1	OK
STIFF1	10	LE1	235,7	35,1	OK
STIFF1	10	LE1	235,9	40,8	OK
STIFF1	10	LE1	235,9	40,8	OK

	f_y [MPa]	f_{tm} [1e-4]
S 235	235,0	500,0

Name	Edge	Size [mm]	Load case	$\sigma_{w,Ed}$ [MPa]	σ_{\perp} [MPa]	UT [%]	Check status
C-flt 1	B-flt 1	6	LE1	260,2	-171,3	72,3	OK
C-flt 1	B-flt 1	6	LE1	335,2	127,8	93,1	OK
C-flt 1	B-web 1	6	LE1	87,7	-43,8	24,4	OK
C-bfl 1	STIFF1	5	LE1	131,9	-95,1	36,7	OK
C-web 1	STIFF1	5	LE1	61,4	4,5	17,0	OK
C-flt 1	STIFF1	5	LE1	263,4	95,7	73,2	OK
C-bfl 1	STIFF1	5	LE1	173,0	-52,7	48,0	OK
C-web 1	STIFF1	5	LE1	44,9	-19,2	12,5	OK
C-flt 1	STIFF1	5	LE1	220,2	140,1	61,2	OK
C-bfl 1	STIFF1	5	LE1	175,9	58,7	48,9	OK
C-web 1	STIFF1	5	LE1	106,0	-15,4	29,4	OK
C-flt 1	STIFF1	5	LE1	228,0	-144,0	63,3	OK
C-bfl 1	STIFF1	5	LE1	140,5	95,1	39,0	OK
C-web 1	STIFF1	5	LE1	45,4	14,1	12,6	OK
C-flt 1	STIFF1	5	LE1	276,9	-91,6	76,9	OK